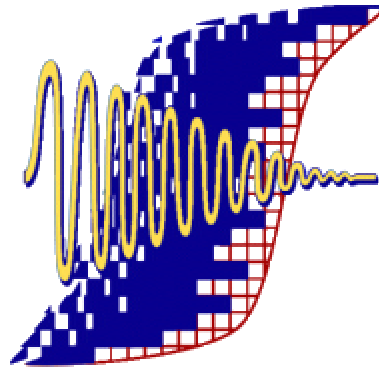


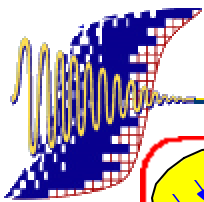
Proof of Concept Investigation of Active Velcro For Smart Attachment Mechanisms



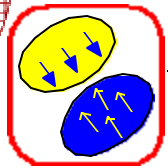
Diann Brei, PhD., Assistant Professor
Joseph Clement, PhD. Pre-Candidate



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Motivation

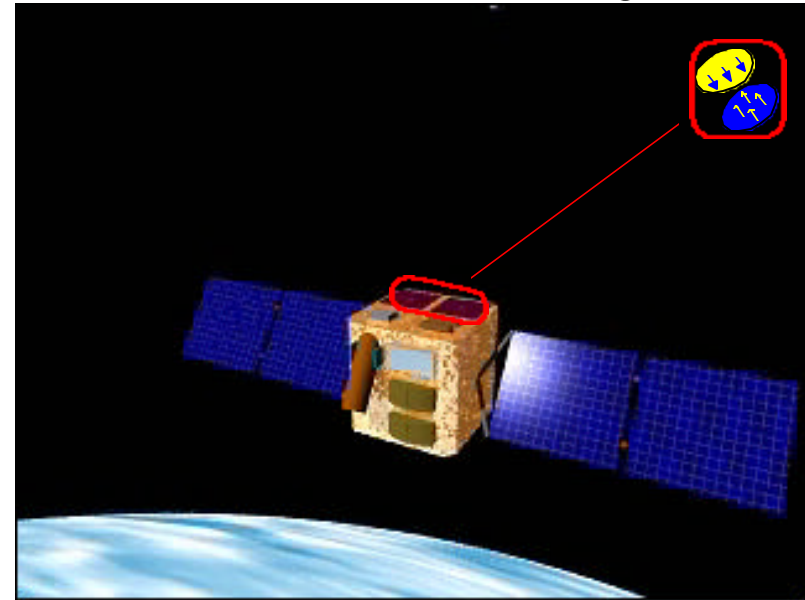


– *Smart Attachment Mechanism*

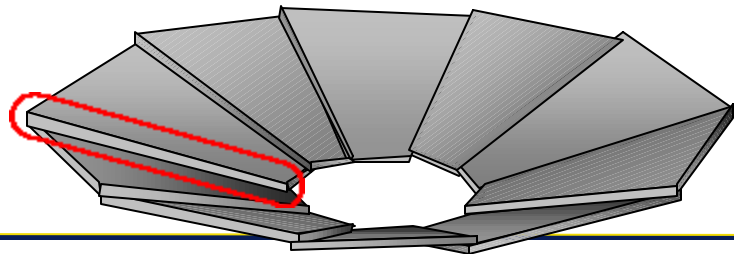
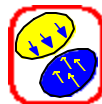
Locomotion along host satellite



Autonomous docking



Shape control for large deployable objects



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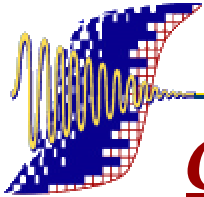
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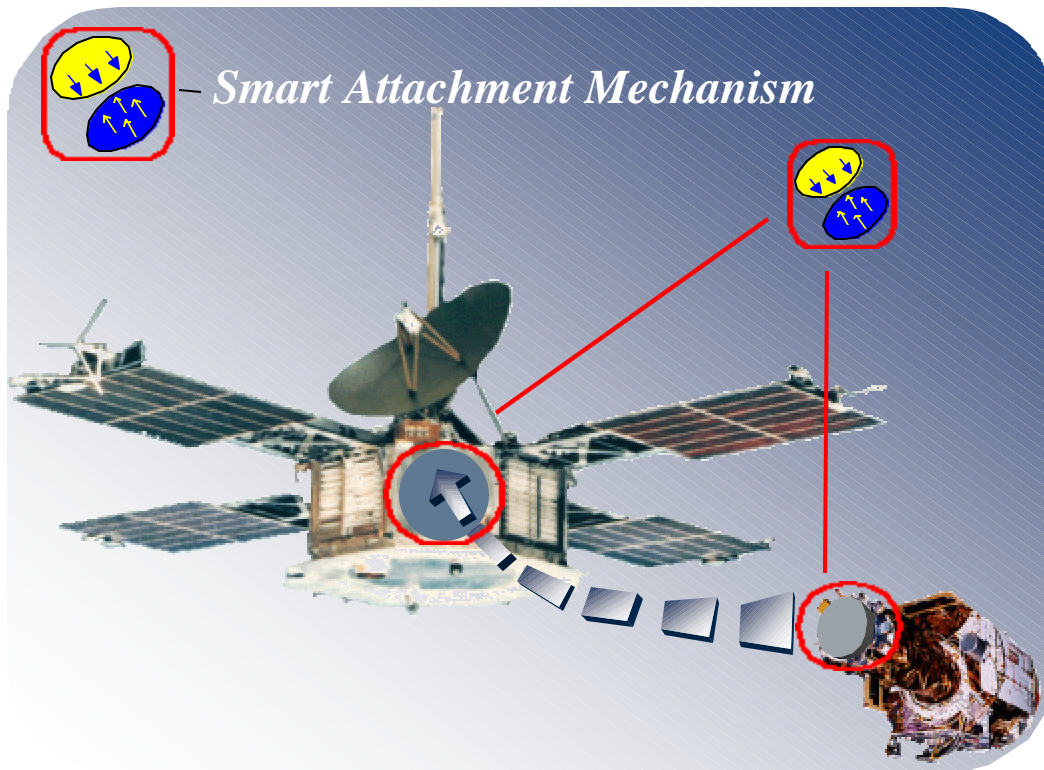
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Active Velcro for Autonomous Satellite Docking



Goal: Develop an active surface capable of autonomously docking two orbiting satellites with precise position and orientation control.



Potential Benefits

Micro-Satellite

- reduced cost / weight
- reduced volume
- reduced complexity (navigation, guidance, homing, etc)

Host Satellite

- mission adaptability
- extended service life
- maintenance/repair facilitation
- module upgrade ability



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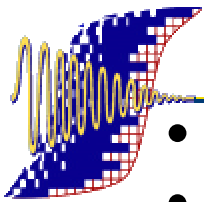
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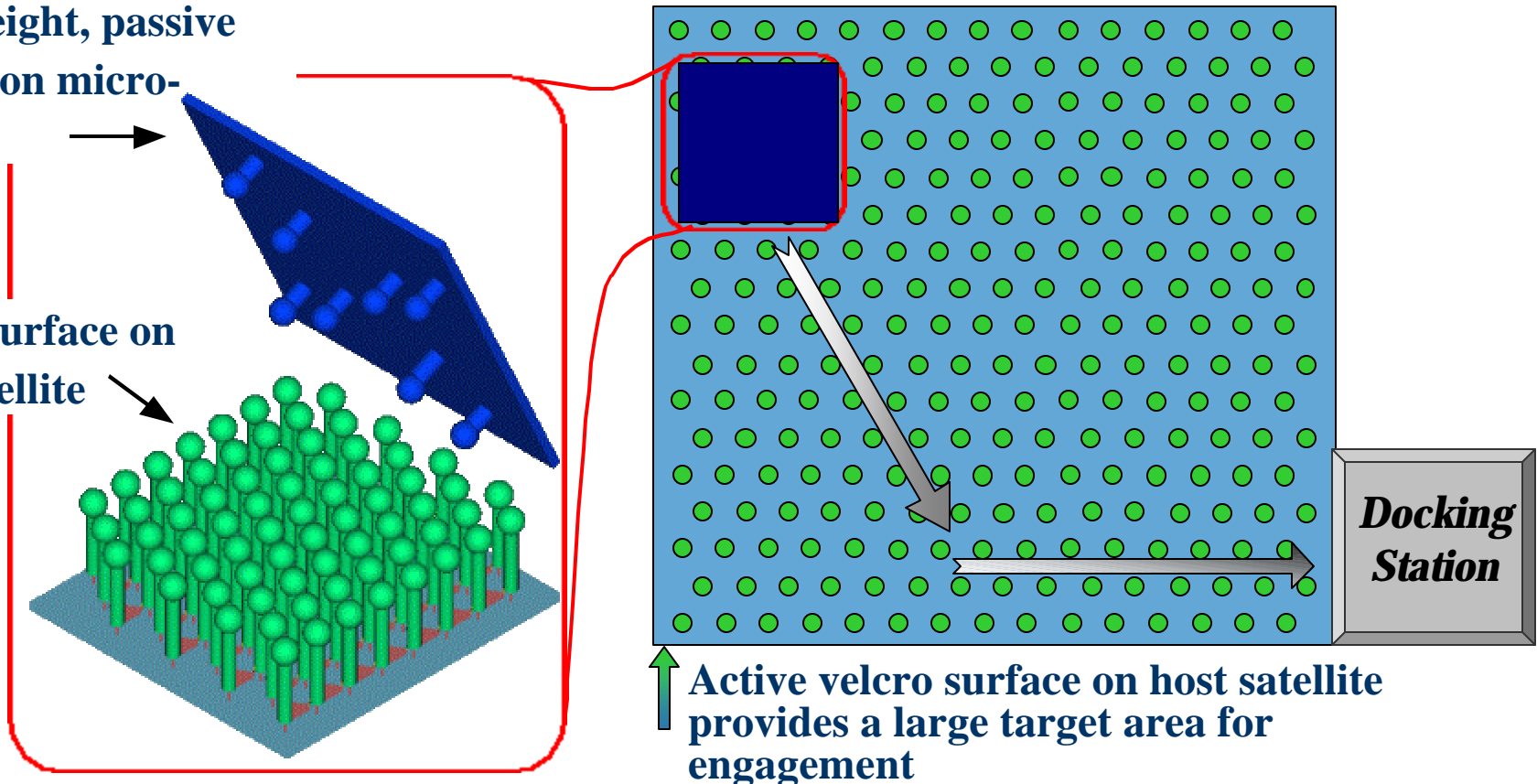


Active Velcro Overview

- The two surfaces passively *latch* on to each other like Velcro in contact.
- The SMA activated prongs on the active velcro surface inch the micro-satellite along the surface of the host satellite to dock.

Lightweight, passive
surface on micro-
satellite

Active surface on
host satellite



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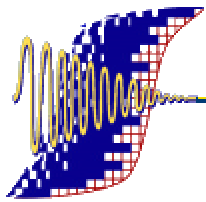
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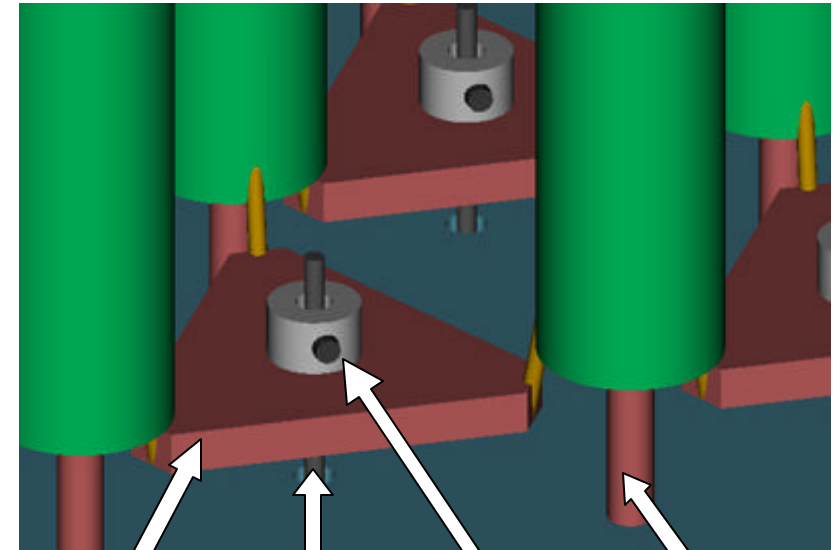
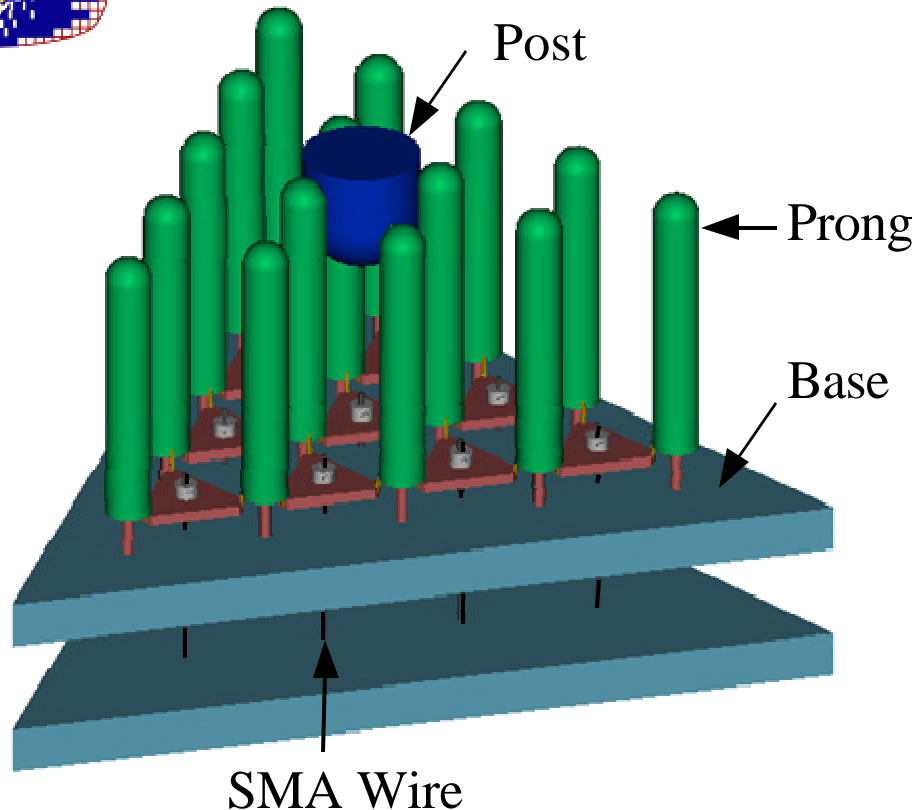
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SMA Actuated Active Velcro Surface*



Actuation Triangle SMA Wire Collet Flexure Tubes

Flexure Mechanism Close-up

*Retention topology omitted for clarity



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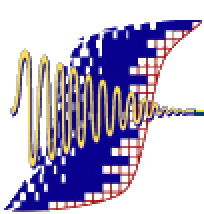
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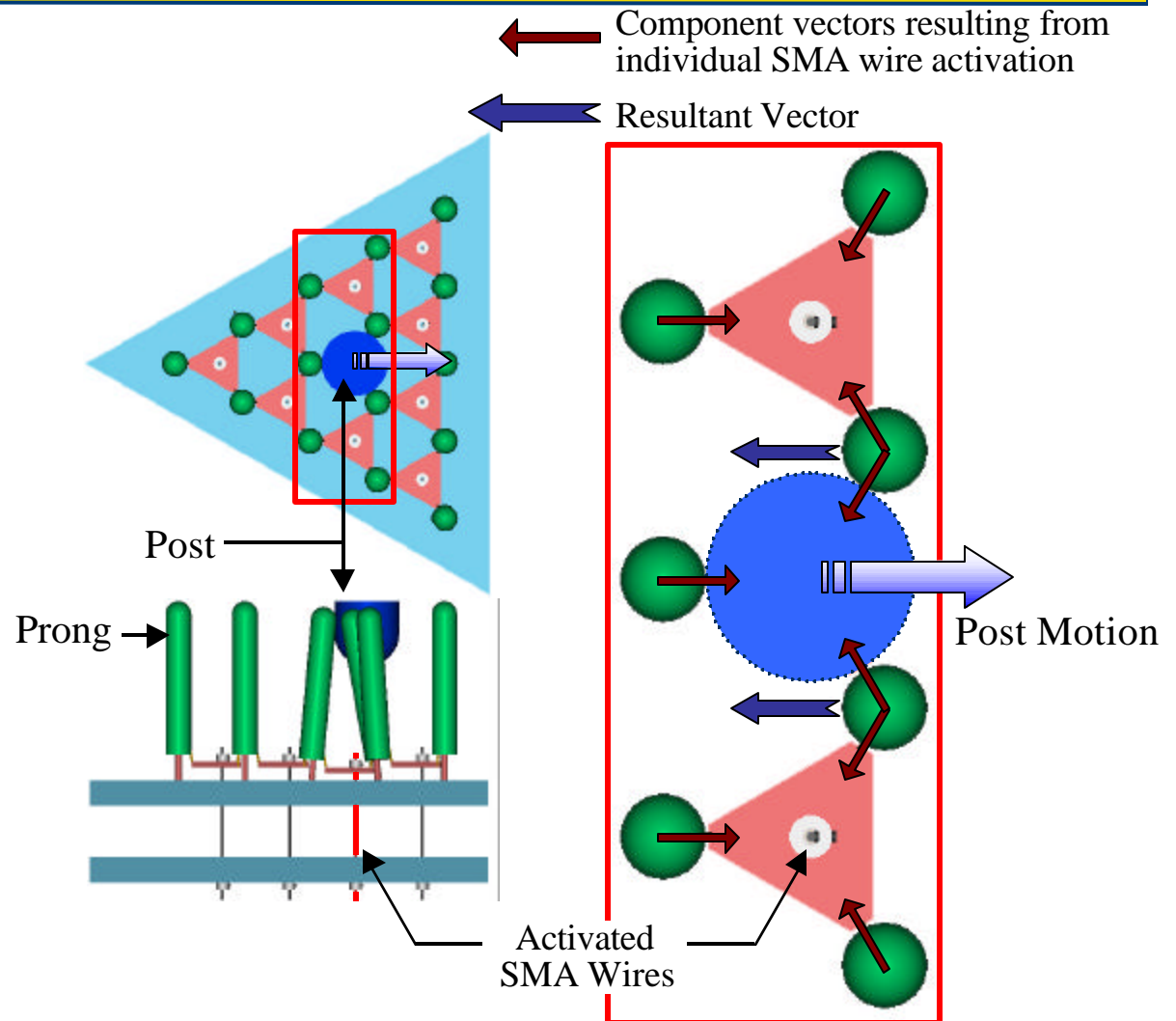
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Local Prong Operation

- Activating a single SMA wire causes the three adjacent prongs to bend inward toward the center of the grouping. ←
- When adjacent SMA wires are activated, the resultant (←) prong motion creates a path for the advancing post. →



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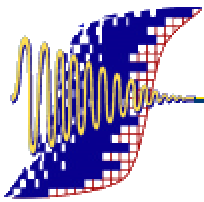
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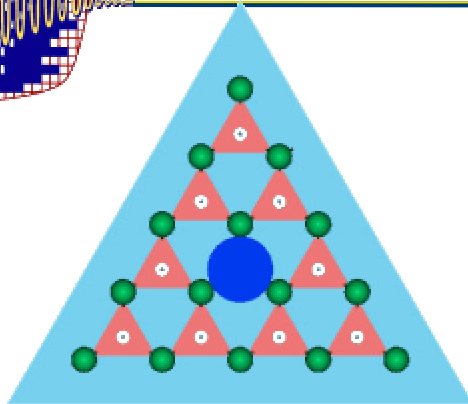
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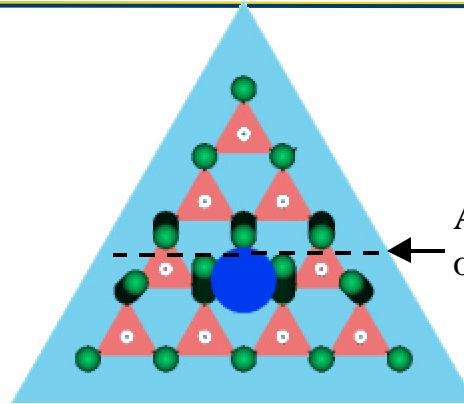
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System Operation

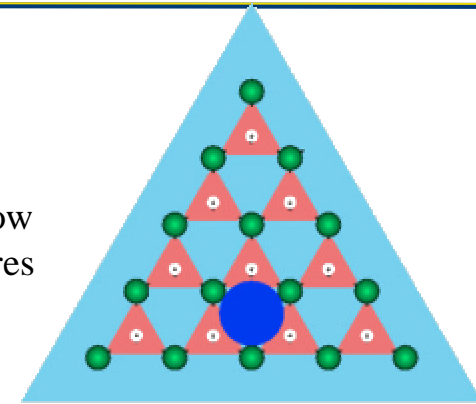


Position 1
Unactivated surface

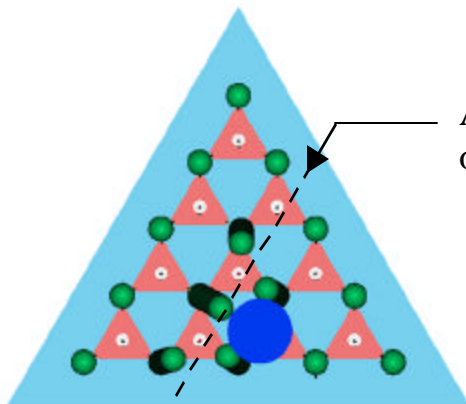


Mid-Translation 1-2
Activated row of SMA wires pushes post into next position

Activated row
of SMA wires

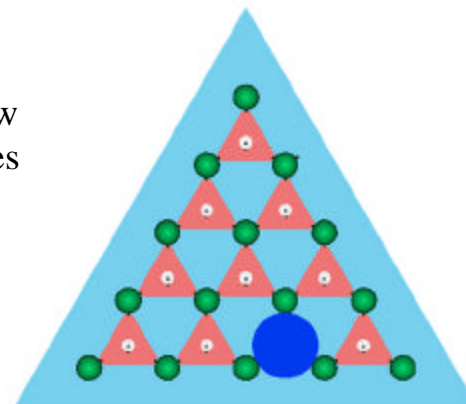


Position 2

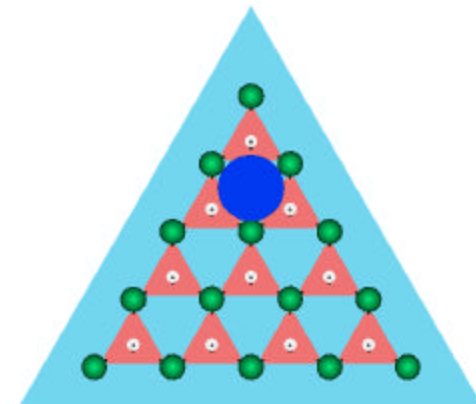


Activated row
of SMA wires

Mid-Translation 2-3
**A different row of SMA wires is activated to advance
post into next position**



Position 3



Simulation



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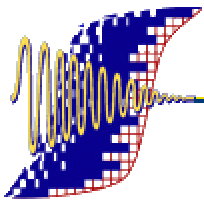
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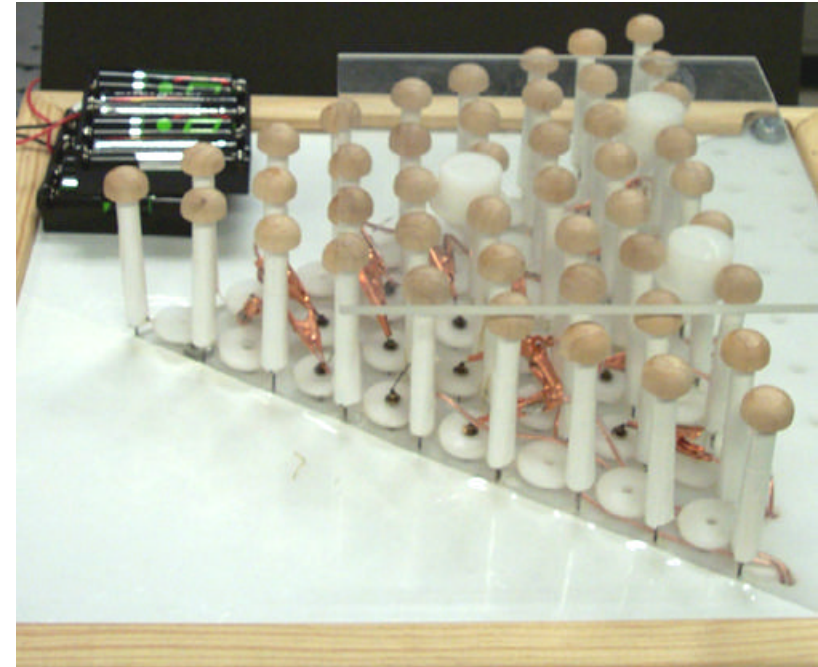
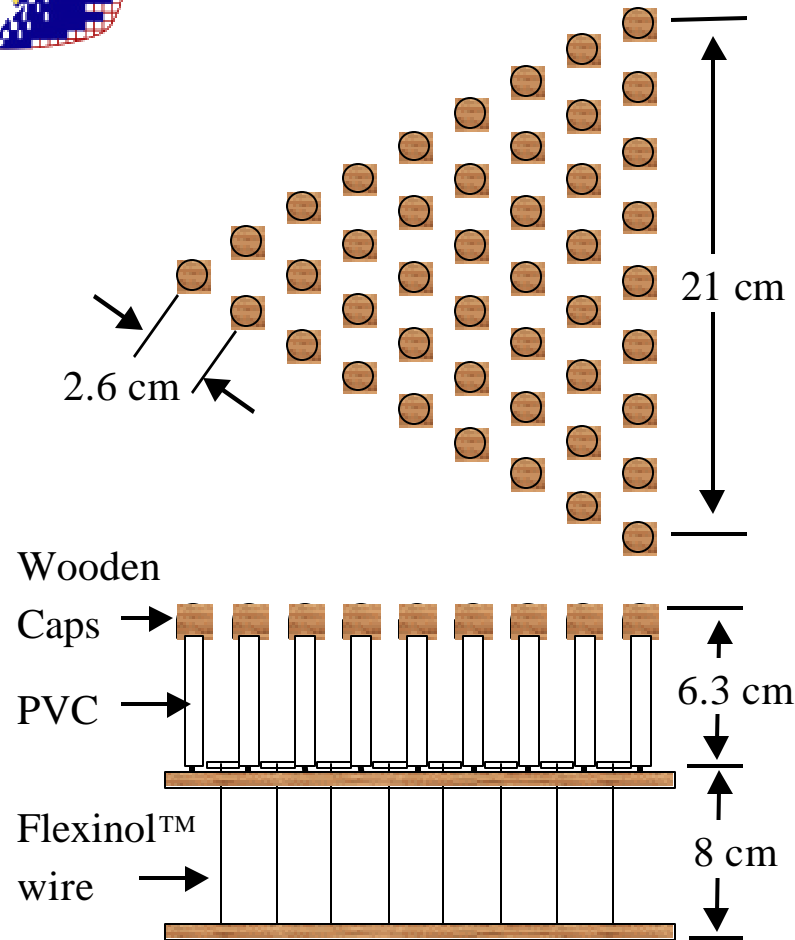
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Large-Scale Prototype



Step Size: 15 mm

Linear Speed: 1.5 - 15 mm/s

Input Power (per step) : 3.3 W

Demonstrated Translation: 1 Post : 8 steps
3 Posts : 2 steps



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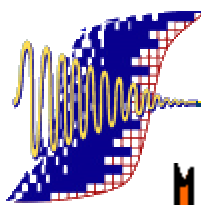
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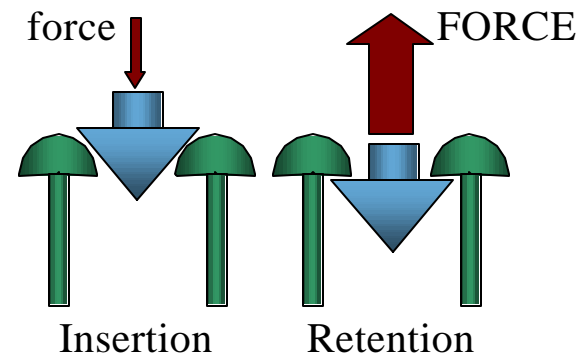
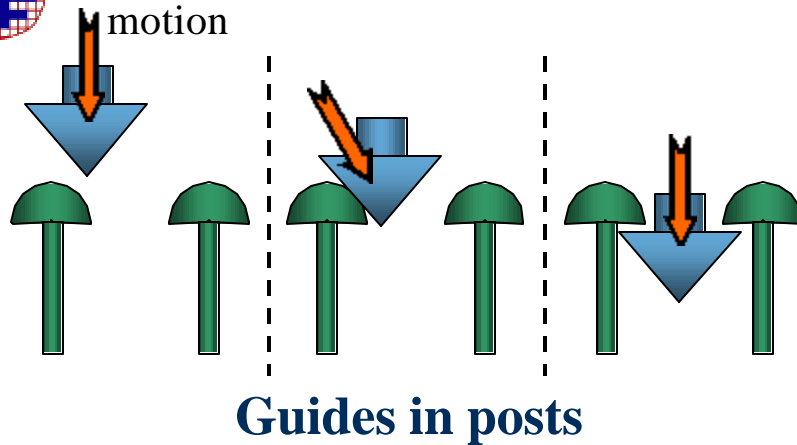
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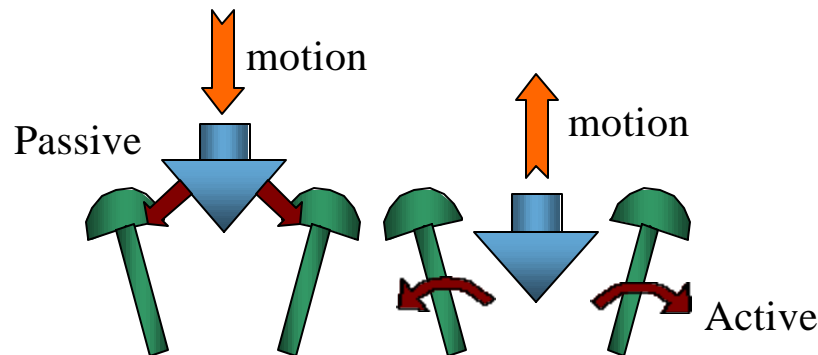
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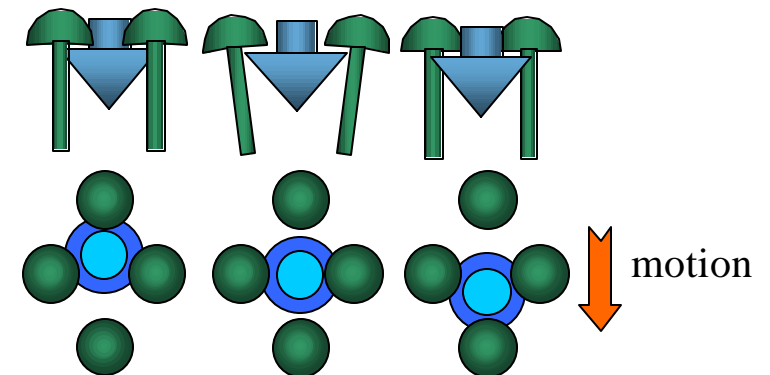
Topology Goals



**Minimum insertion force /
Maximum retention force**

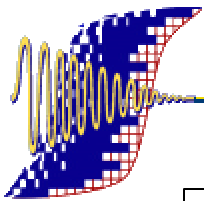


**Passive engagement /
Active disengagement**



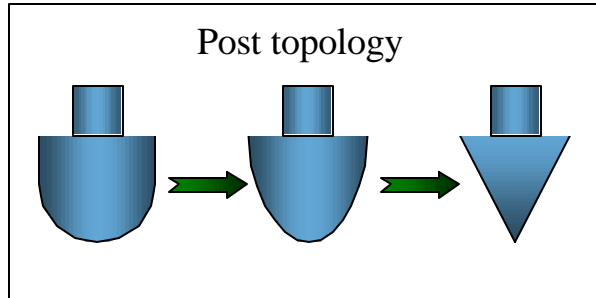
Enables motion (translation, rotation)



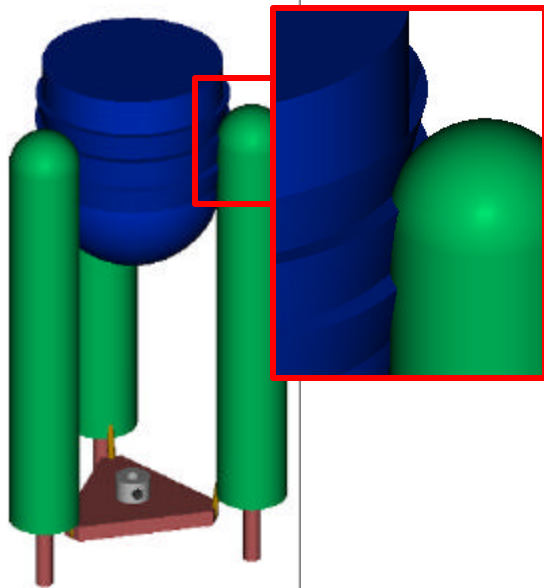
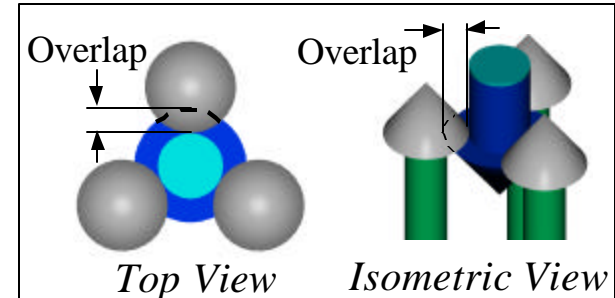


Retention Topology Concepts

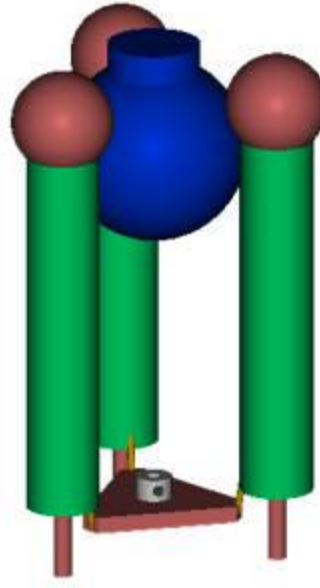
Guidance



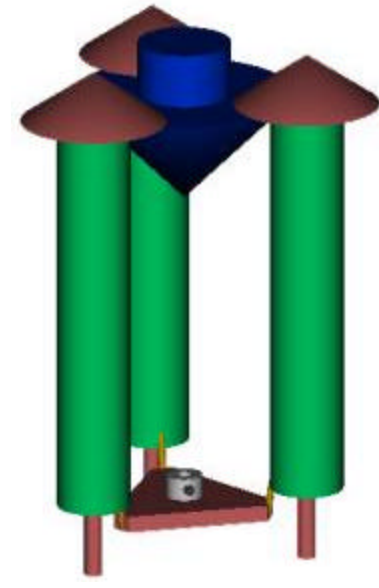
Retention



Burred Surfaces



Spherical Ends



Conical Ends



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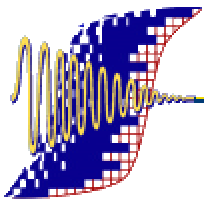
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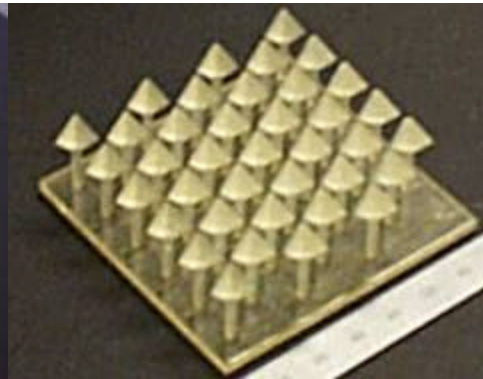
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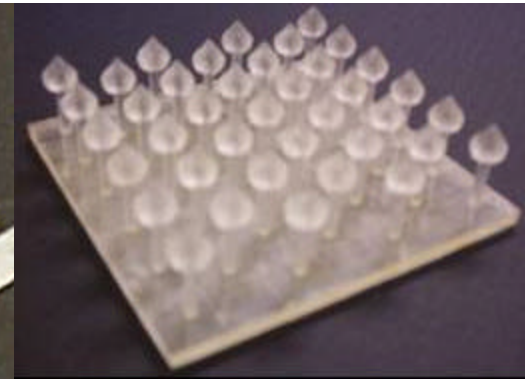
Stereolithography Topology Prototypes



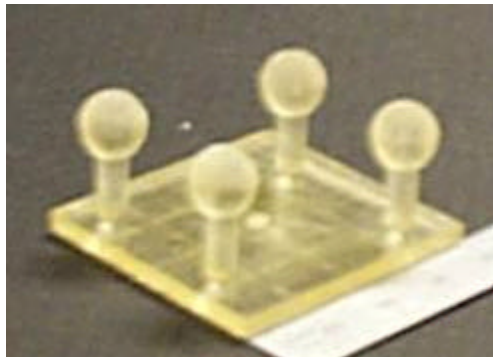
Sphere Prongs



Tetrahedron Prongs



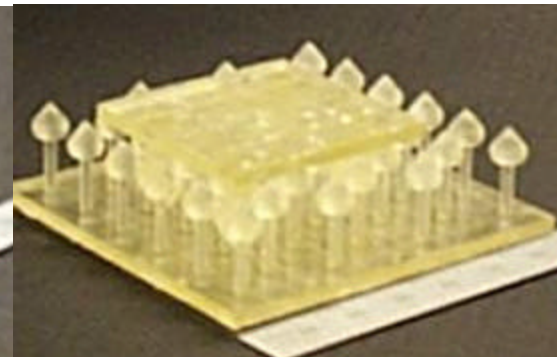
Bulb Prongs



**Sphere/Tetrahedron
Posts**



Bulb Post



Inserted Configuration



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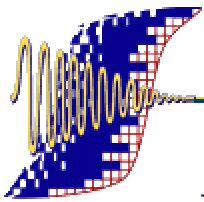
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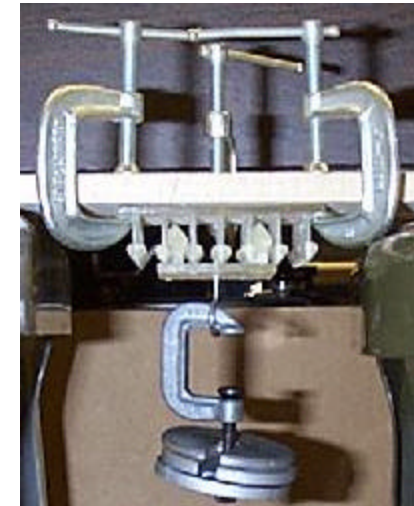
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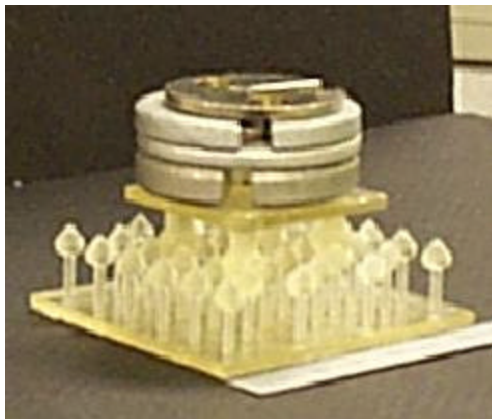
Topology Experimental Study Results

- **Lessons Learned:**

- Sphere: provides low engagement forces at the cost of reduced retention force
- Tetrahedron: excellent engagement capabilities but poor retention force
- Bulb: provides good balance of engagement and retention capabilities



Disengagement



Engagement

	Engagement Force (N)	Disengagement Force (N)
Sphere	4.7	6.9
Tetrahedron	2.9 / 22.6	1.8 / 9.9
Bulb	8.6	10.1



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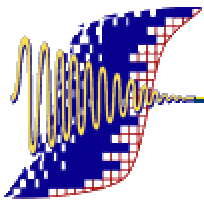
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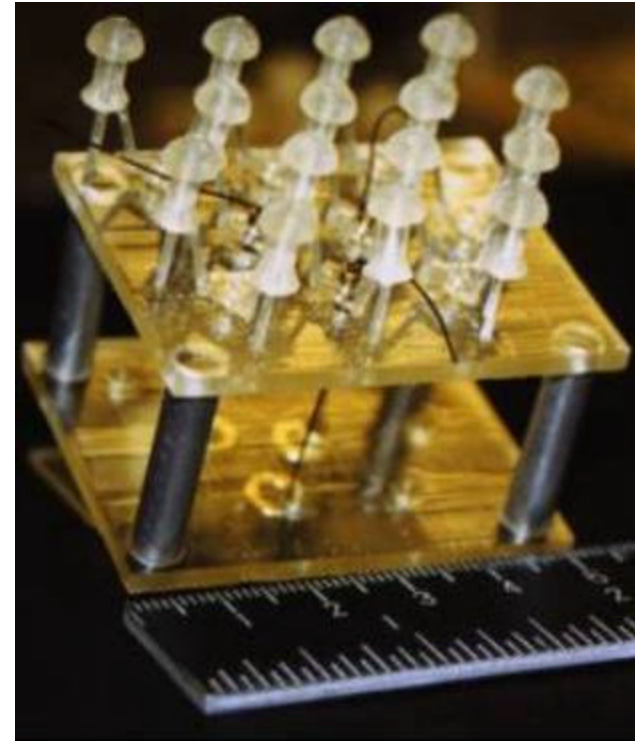
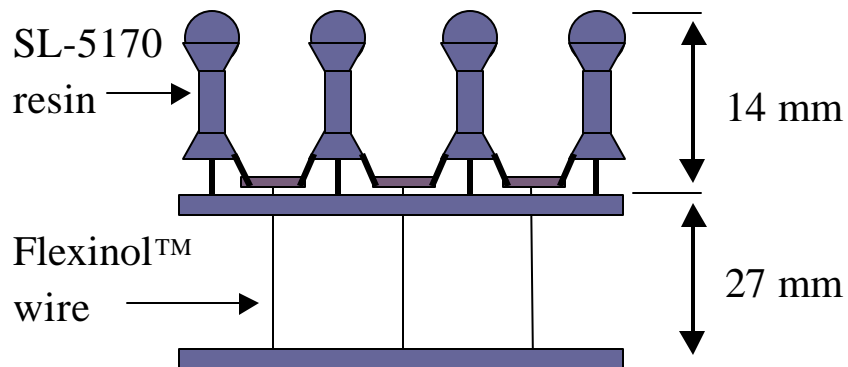
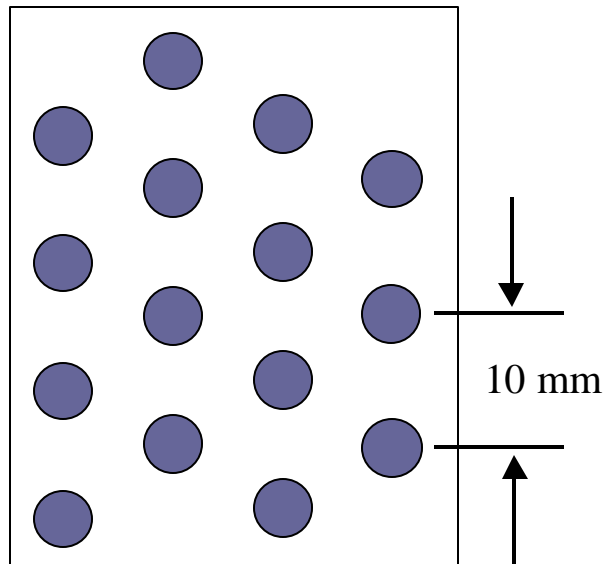
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Small-Scale Prototype



Step Size: ~ 6 mm
Linear Speed: 0.6 - 6 mm/s
Input Power (per step): 2.3 W
Demonstrated Translation



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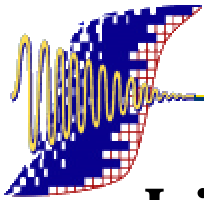
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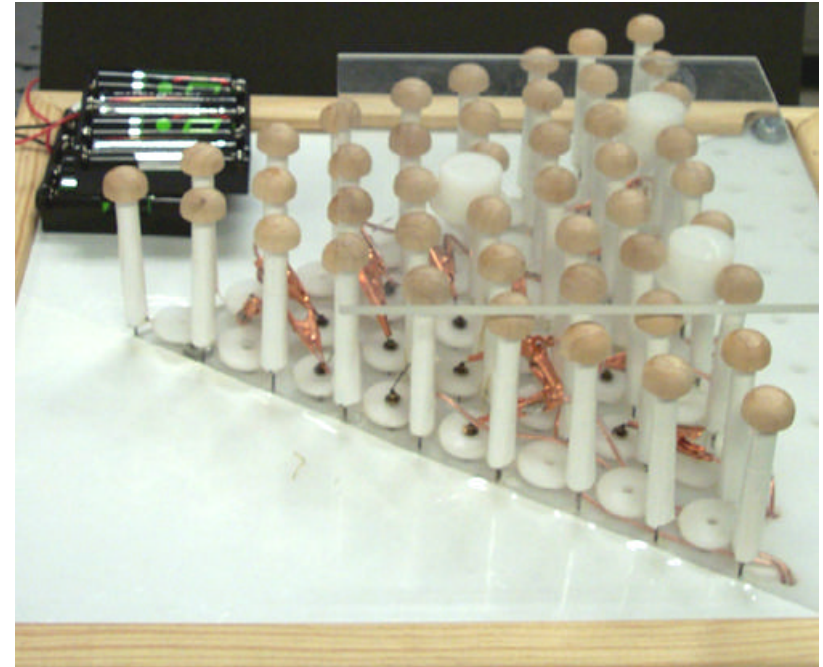
13



Accomplishments

Initial concept generation

- Several unique concepts were generated
- An active material selection process was performed
- Designs were initially modeled based on buckling loads, actuator force/deflection criteria, power requirements and failure mechanisms
- Down selection to final design with developed evaluation metrics



A simple large-scale proof-of-concept prototype

- Constructed with off-the-shelf components (~ 11x11x14 cm)
- ***In initial tests, prototype demonstrated planar translation***



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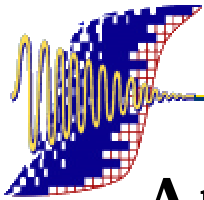
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Accomplishments Continued

A topology study was conducted

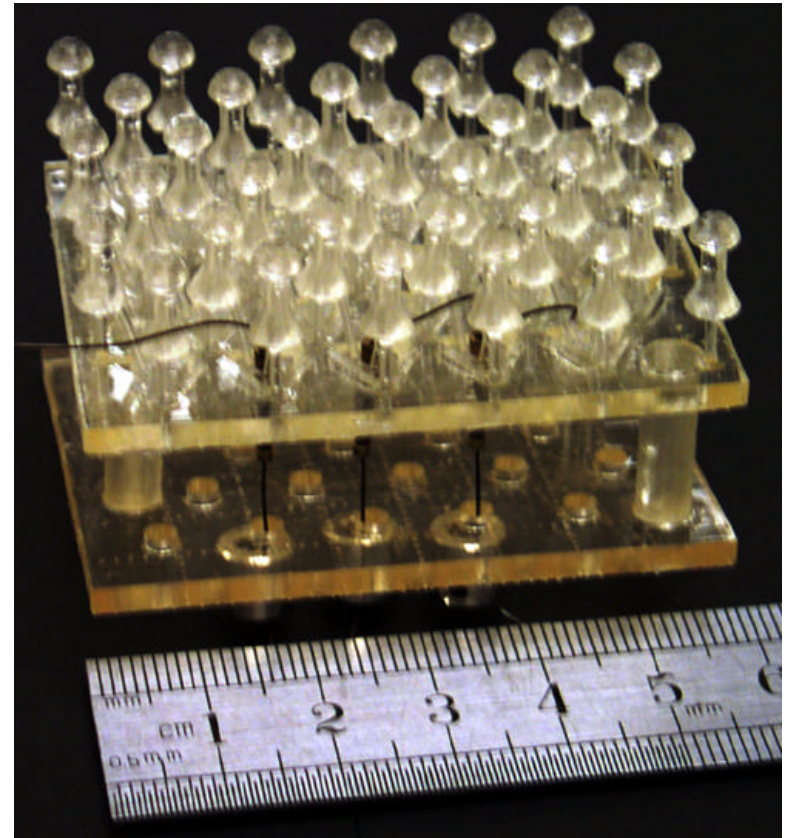
- Engagement, retention and required translation forces may be tailored independently
- Small changes in the surface finish (burrs) have been demonstrated to significantly increase retention forces

Reduced-scale stereolithography prototypes have been constructed

- Varied engagement and retention forces through connection topology alterations
- One piece flexure mechanism and structural backbone

- ***In initial tests, prototype demonstrated planar translation***

Patent filed and pending



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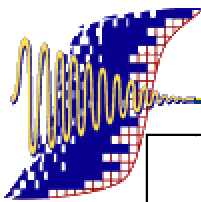
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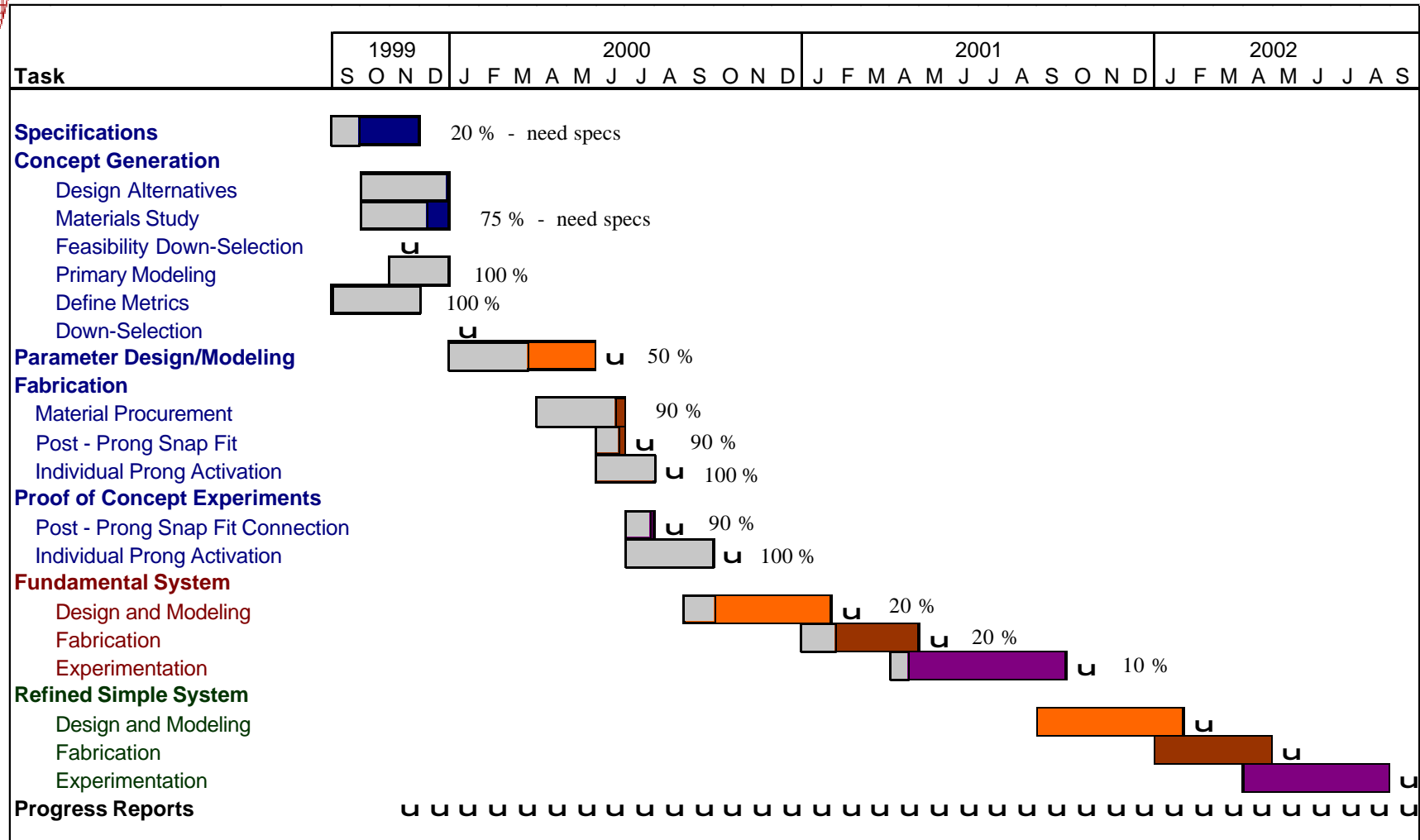
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Schedule



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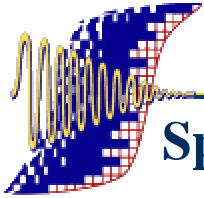
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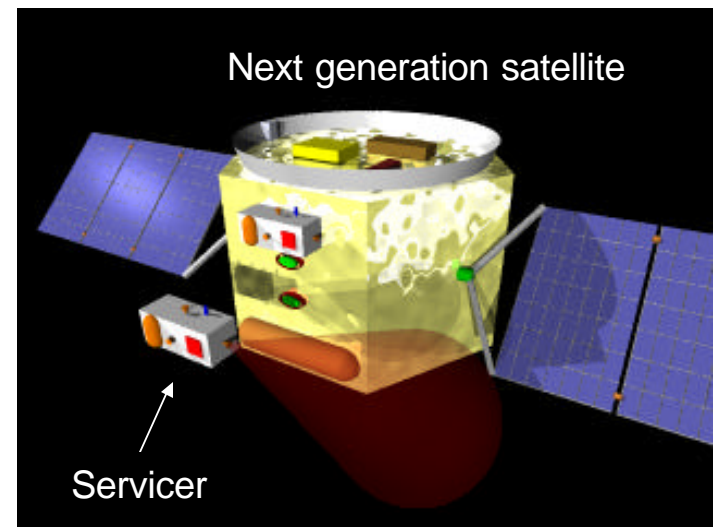
Technology Transfer and Potential Customers

Space Structures

- Preliminary discussions with Honeywell on commercialization and licensing initiated
- XSS-11 is being explored as a possible experimental satellite platform to test servicing capabilities on the experimental satellite
- Potential use for course-accuracy docking on Orbital express to save on guidance and control complexity when plugging in orbital replacement units

Potential Markets

- Manufacturing
 - assembly and precision connections
 - precision shaping and placement
 - fixturing and alignment
 - reconfigurable tooling
- Telecommunications and Optics
 - fiber optic placement, connection, and alignment
 - precision lens/mirror alignment and shape control
- Medicine
 - bio attachments
 - sutures
 - alignment of prosthetics
 - assistive surgical tools



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University of Michigan

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